# DETERMINATION OF POROSITY LEVEL OF OUT OF AUTOCLAVE CARBON/EPOXY ADVANCED COMPOSITES USING ULTRASONIC C-SCAN (PULSE-ECHO METHOD)

### MOHD FADZLEE BIN ZAINAL ABIDIN

UNIVERSITI SAINS MALAYSIA 2017

## DETERMINATION OF POROSITY LEVEL OF OUT OF AUTOCLAVE CARBON/EPOXY ADVANCED COMPOSITES USING ULTRASONIC C-SCAN (PULSE-ECHO METHOD)

by

### MOHD FADZLEE BIN ZAINAL ABIDIN

Thesis submitted in fulfilment of the requirements for the degree of Master of Science

### ACKNOWLEDGEMENTS

I would like to thank Allah the Almighty for His blessings. I take this opportunity to sincerely thank my supervisor Dr Elmi bin Abu Bakar and co-supervisor Associate Professor Dr Roslan bin Ahmad for their constant source of inspiration, support and bailing me out in times of difficulty and tricky situations during the experimental phase. My thanks and gratitude to the ex-supervisor Associate Professor Dr Abdul Rahim Othman for giving me the opportunities also the group of doing the Out of Autoclave project. To my colleague at AMREC, SIRIM Berhad; Suhaimi Wahap, Mohamad Afizoll Mohamad Pisol, Mawarni Mohamad and Dr Mohd Asri Selamat - I'm truly in your debts.

Last, but far from least, I wish to express my gratitude to my family and especially to my wife, Shirin binti Ibrahim and children Mohammad Aishdanni, Anna Aisya and Alisha Alya for their moral support, encouragement and love.

### TABLE OF CONTENT

			Page
ACI	KNOW	LEDGEMENT	ii
TAI	BLE O	F CONTENT	iii
LIS	T OF F	TIGURES	vi
LIS	T OF T	ABLES	xi
LIS	T OF A	ABBREVIATIONS	xii
LIS	I OF S	YMBOLS	xiv
ABS	STRAK		XV
ABS	STRAC	T	xvi
CH	APTER	ONE: INTRODUCTION	
1.1	Backg	round	1
1.2	Proble	em Statement	2
1.3	Objec	tives	3
1.4	Scope	of Study	4
CH	APTER	R TWO: LITERATURE REVIEW	
2.1			
2.2	Ultras	onic C-scan testing	6
	2.2.1	Pulse echo	7
	2.2.2	Attenuation measurement	8
	2.2.3	Defect detection using ultrasonic C-scan for homogeneous materials.	8
	2.2.4	Defect detection using ultrasonic C-scan for heterogeneous materials.	12
		2.2.4(a) Ceramic Matrix Composite (CMC)	12
		2.2.4(b) Metal Matrix Composite (MMC)	14
		2.2.4(c) Polymer Matrix Composite (PMC)	16
	2.2.5	Analysis of ultrasonic C-scan for composite laminates	21
		2.2.5(a) Images processing	24
		2.2.5(b) Signal processing	30

		2.2.5(c) Combination of signal/image processing and its contribution.	34
2.3	Summ	nary of literature review	37
	2.3.1	Signal processing	37
	2.3.2	Image processing	38
	2.3.3	Combination of signal and image processing.	39
СН	<b>A PTER</b>	R THREE: MATERIAL AND METHOD	
3.1		ration of carbon/epoxy composite laminate.	40
3.1	3.1.1	Prepreg materials data	42
		Mould preparation	43
	3.1.2	3.1.2(a) Mould cleaning.	43
		3.1.2(b) Waxing and buffing	43
3.2	Prena	ration of prepreg material.	44
3.2	3.2.1	Defrost process (thawing)	44
		Prepreg sectioning process	44
3.3		ding process	44
3.3	3.3.1		44
	3.3.2		45
	3.3.2	3.3.2(a) Summary of vacuum bagging process	45
	3.3.3	Vacuum process	46
	3.3.4	Curing process	47
	3.3.5	Demoulding process	47
3.4		ration of reference sample	48
3.4	_	lishment of ultrasonic C-scan testing method.	48
3.6		processing	49
3.7	•	volume	49
3.7	3.7.1		50
	3.7.1	Theoretical density calculation  Void content	
2.0			50
3.8	mage	e analysis using Matlab	51
CH	APTER	R FOUR: RESULT AND DISCUSSION	
4.1	Ultras	sonic C-scan	52

4.2	Analy	sis data usi	ing Matlab	61
	4.2.1	Time of f	Hight (TOF)	65
	4.2.2	Amplitud	le (AMP)	79
	4.2.3	Void volu	ume	86
		4.2.3(a)	Theoretical density calculation	87
		4.2.3(b)	Void Content	87
4.3	Image	analysis u	sing Matlab	88
CH	APTER	R FIVE: C	ONCLUSION AND RECOMMENDATION	
5.1	Concl	usion		96
5.2	Recon	nmendation	n for future work	98
REI	EREN	CES		99

### LIST OF FIGURES

		Page
Figure 2.1	Quickstep QSE250 machine and curing chamber	6
Figure 2.2	Ultrasonic testing principle	7
Figure 2.3	Comparative results of three different titanium specimens by (a) Ultrasonic C-Scan, and (b) VT	9
Figure 2.4	Calibration sample	9
Figure 2.5	Gated C-scan image of gun barrel showing the defect area & the corresponding B-scan.	10
Figure 2.6	C-Scan image with the 10MHz transducer of fatigued specimen with various fatigue life (a) 0%, (b) 55%, (c) 75%, (d) 85% and (e) Average attenuation measured with 10 MHz and 20 MHz transducer	11
Figure 2.7	Correlation between defect size and echo amplitude in steel and grey cast iron.	12
Figure 2.8	Schematic diagram of theoretical A-scan showing reflections from interfaces within a multiple layered-structure. For clarity, the waveform shown includes only the components of the waveform due to the Ceramic/Matrix and the Matrix/Elastomer interfaces.	13
Figure 2.9	a) B-scan of Ceramic-side inspection showing the different resonances for different tile materials, b) FFT-filtered C-scan tuned to examine the lower-performance tile regions.	14
Figure 2.10	C-Scan image of reference block recorded with ScanMasters LS 100 immersion system.	15
Figure 2.11	80 MHz C-scans taken from different ply reflections. Coordinates in mm. The darker regions represent a stronger signal on these images. Yellow corresponds approximately to the noise threshold	16
Figure 2.12	Multiple reflections in the signal measured at the position over the delamination type defect.	18
Figure 2.13	Pitch-catch measurement on composite laminate using miniature potted angle beam transducers.	19
Figure 2.14	C-scan pulse-echo ultrasonic image of a specimen using 2.25, 5 and 10 MHz sensors	25
Figure 2.15	Ultrasonic C-Scan after 40J impact.	26

Figure 2.16	B-scan after 40J impact.	26
Figure 2.17	<ul><li>a) Reference image b) C-scan image c) Difference image</li><li>d) Potential defect</li></ul>	27
Figure 2.18	The drawing of the measured sample in situ (CFRP sample)	28
Figure 2.19	Microphotograph of a carbon-epoxy material with a localized porous zone (8 central plies).	32
Figure 3.1	Bagging construction	45
Figure 3.2	a) Vacuum bagging b) Walk-in curing oven	46
Figure 3.3	Composite panels.	47
Figure 3.4	NDT Ultrasonic C-Scan	49
Figure 3.5	Image analysis step using Matlab®	51
Figure 4.1	Sample Ref TOF	53
Figure 4.2	Sample Ref AMP	53
Figure 4.3	Sample A1 TOF	53
Figure 4.4	Sample A1 AMP	53
Figure 4.5	Sample A2, TOF	54
Figure 4.6	Sample A2, AMP	54
Figure 4.7	Sample A3, TOF	55
Figure 4.8	Sample A3, AMP	55
Figure 4.9	Sample O1, TOF	55
Figure 4.10	Sample O1, AMP	56
Figure 4.11	Sample O2, TOF	56
Figure 4.12	Sample O2, AMP	56
Figure 4.13	Sample O3, TOF	57
Figure 4.14	Sample O3, AMP	58
Figure 4.15	Sample O4, TOF	58
Figure 4.16	Sample O4, AMP	58

Figure 4.17	Sample O5, TOF	59
Figure 4.18	Sample O5, AMP	59
Figure 4.19	Comparison of thickness.	59
Figure 4.20	Surface using PTFE	60
Figure 4.21	Surface not using PTFE	60
Figure 4.22	Laminate construction (a) and inverted construction during scanning (b)	62
Figure 4.23	Command of average thickness.	62
Figure 4.24	Density of samples	65
Figure 4.25	Command of fixed thickness (a) reference sample (b) sample A2.	65
Figure 4.26	Defect in plies thickness	66
Figure 4.27	A1 image using Matlab	66
Figure 4.28	3D image for A1	67
Figure 4.29	A2 TOF image using Matlab	68
Figure 4.30	Replotted 3D image for A2	68
Figure 4.31	Replotted 3D image for A3	69
Figure 4.32	Dry spot area of sample A2 from C-scan image	70
Figure 4.33	Defect in sample A3 from C-scan	71
Figure 4.34	Scanned surface image (no defect the surface)	71
Figure 4.35	Closed up porous surface	71
Figure 4.36	Sample O2 replotted using Matlab	73
Figure 4.37	C-scan image shows that the defect at the edges of sample and thickness at the centre of sample O3	73
Figure 4.38	Replotted 3D image for O1	74
Figure 4.39	Replotted 3D image for O2	74
Figure 4.40	Defected area on surface (dry spot)	74
Figure 4.41	Replotted 3D image for O3	75

Figure 4.42	Replotted 3D image for O4	76
Figure 4.43	Replotted 2D images for O4	77
Figure 4.44	Captured image of O4 surface	77
Figure 4.45	C-scan image of sample O5	78
Figure 4.46	O5 TOF image using Matlab	78
Figure 4.47	3D image of O5 using Matlab	79
Figure 4.48	Command for AMP	79
Figure 4.49	AMP images (a) A1 (b) A2 (c) A3	82
Figure 4.50	Replotted O1 AMP	83
Figure 4.51	Replotted O2 AMP	84
Figure 4.52	Replotted O3 AMP	84
Figure 4.53	Replotted O4 AMP	85
Figure 4.54	Replotted O5 AMP	86
Figure 4.55	Actual density of samples	87
Figure 4.56	Comparison between destructive and image processing	90
Figure 4.57	Matched image of A1	91
Figure 4.58	Matched image of A2	91
Figure 4.59	Matched image of A3	92
Figure 4.60	Matched image of O1	93
Figure 4.61	Matched image of O2	93
Figure 4.62	Matched image of O3	94
Figure 4.63	Matched image of O4	94
Figure 4 64	Matched image of O5	95

### LIST OF TABLES

		Page
Table 2.1	Ultrasonic velocity values of samples by 4 MHz probe	14
Table 2.2	Summary of PMC ultrasonic NDT	20
Table 2.3	Summary of image processing ultrasonic NDT	30
Table 2.4	Summary of signal processing ultrasonic NDT	33
Table 3.1	Number of ply involved reflect with prepreg used	41
Table 3.2	List of supporting tools and equipment involved in fabrication process	41
Table 3.3	Prepreg properties	42
Table 3.4	Bagging construction	46
Table 3.5	Sample Identification	47
Table 4.1	Scanning parameter	52
Table 4.2	Average thickness	63
Table 4.3	Calculated percentage of different thickness	64
Table 4.4	Average amplitude	80
Table 4.5	Average thickness and amplitude	80
Table 4.6	Maximum and Minimum Amplitude	81
Table 4.7	Theoretical density	87
Table 4.8	Void content	88
Table 4.9	Percentage of intensity and matching	88
Table 4.10	Defect comparison	89

### LIST OF ABBREVIATIONS

OOA Out of Autoclave

MMC Metal Matrix Composite

CMC Ceramic Matrix Composite

PMC Polymer Matrix Composite

UT Ultrasonic Testing

MPI Magnetic Particle Inspection

NDT Non-Destructive Testing

2D Two Dimensional

3D Three Dimensional

dB Decibel

TOF Time of Flight

AMP Amplitude

PTFE Polytetrafloroethylene

CFRP Carbon Fibre Reinforced Plastic

GFRP Glass Fibre Reinforced Plastic

TiMMC Titanium Metal Matrix Composite

TIRP Thickness Free Reflector Plate

BSE Back Surface Echo

SNR Signal to Noise Ratio

UEPM Ultrasonic Energy Propagation Movie

UPGMA Unweighted Pair Group Method with Arithmetic Mean

FD Fractal Dimension

ATR Automatic Target Recognition

PEEK Polyetheretherketone

BSS Blind Signal Separation

SDAC Spectral Distance Amplitude Correction

WPT Wavelet Packet Transform

SEM Scanning Electron Microscopy

FFT Fast Fourier Transforms

RGB Red, Green, Blue

ASTM American Standard Test Method

### LIST OF SYMBOLS

MHz Megahertz kHz Kilohertz MHz Megahertz

### TAHAP KELIANGAN OLEH PENGUJIAN PENGIMBASAN ULTRASONIK (KAEDAH GEMA DENYUT) UNTUK LAMINAT KARBON/EPOKSI DIPERBUAT MENGGUNAKAN KAEDAH LUAR AUTOKLAF

### **ABSTRAK**

Teknik pemeriksaan menggunakan ultrasonik C-scan adalah yang paling sinonim dalam industri pembuatan bahan laminat komposit termaju. Tidak ada piawaian yang boleh diterima pakai secara meluas atau prosedur untuk pemeriksaan C-scan ultrasonik untuk bahan yang berasaskan komposit termaju yang mana boleh mengakibatkan kebolehpercayaan dan kebolehkesanan kaedah ini telah terhad. Pengkajian tentang kaedah dan pengujian bahan komposit termaju telah dijalankan sebagai tindakbalas kepada masalah di atas. Kajian ini telah menghasilkan tiga prosedur untuk mengendalikan kelengkapan C-scan melalui kaedah denyut gema, penghasilan panel rujukan yang mana panel rujukan ini adalah dihasilkan melalui kaedah yang diperakui kualitinya dan melakukan analisis lanjutan dengan penggunaan perisian seperti Matlab yang bertujuan untuk menganalisa kecacatan dalam bentuk imej 3 dimensi. Tesis ini memberikan gambaran keseluruhan objektif dan aktiviti penyelidikan yang telah dijalankan. Sebagai penemuan, penilaian antara ujian musnah dan tanpa musnah (pemprosesan imej) menunjukan kebolehbandingan antara satu dengan yang lain. Sampel O1 adalah sampel yang terbaik dikalangan bahan OOA disediakan dalam experimen ini dengan menghasilkan kecacatan samada keliangan ataupun lompang dengan peratusan yang rendah. Ianya melibatkan pemampatan vakum dengan menggunakan teknik beg vakum yang normal dan dibantu dengan dua (2) sumber vakum dengan kehadiran filem polytetrafluoroethylene yang ditempatkan pada permukaan acuan bagi melancarkan perjalanan resin dipermukaan acuan semasa proses penyediaan sampel. Analisis selanjutnya perlulah dilakukan dalam usaha menilai secara kualitatif.

## DETERMINATION OF POROSITY LEVEL OF OUT OF AUTOCLAVE CARBON/EPOXY ADVANCED COMPOSITE USING ULTRASONIC C-SCAN (PULSE-ECHO METHOD).

### **ABSTRACT**

Various inspection techniques are available for inspection and quality control in monitoring services of composite structures, which ultrasonic inspection techniques C-scan is the most widely used in industrial manufacturing advanced composite materials. At this time, there are no standards that can be applied universally or procedures for inspection ultrasonic C-scan for advanced composite based materials which may cause the reliability and traceability of these methods has been limited. In response to this need, the study of methods and testing of advanced composite materials has been carried out. This assessment has produced three procedures 1. to operate the equipment C-scan through the pulse echo method, 2. producing the reference panel which reference panel is produced by a method recognized for its quality and usability of advanced composite panel produced aims to simulate defects. 3. Perform advanced analysis with use software such as Matlab<sup>®</sup>, which aims to analyze the defects in the form of three-dimensional images. This thesis gives an overview of the objectives and research activities to be carried out. As a finding, evaluation performance between destructive and image processing technique shown that comparable results between each other. Sample indicated O1 was determined as a best sample among others with lowest defect (i.e. porosity and void) percentage. It involved in no debulking process with normal vacuum bagging technique assisted by two (2) vacuum sources with polytetrafluoroethylene film on mould surface to enhance flow during fabrication. Supporting analysis should be carried out in order to analyze qualitatively.

### **CHAPTER ONE**

### INTRODUCTION

### 1.1 Background

The latest generation of prepreg out autoclave has been introduced by prepreg material producers, this latest development of prepregs has shown that it is possible to produce quality autoclave parts using vacuum bag assembly. Without the use of autoclave, the prepreg material developed can reduce the cost of acquisitions and operations, and is compatible with various types of low cost curing process, including conventional ovens, heating blankets, and heated appliances (R.Harshe, 2015).

The use of prepreg out of autoclave (OOA) does not accelerate the production process of advanced composite products. Due air trapped removing during processing, is it a time-dependent process. The curing cycle of OOA technique is usually longer. Typically, after completely conducted debulking process, vacuum must be held for a longer period before starting the curing process. The duration of this vacuum holding depends on the size and complexity, from as low as 4 hours for 0.4 m² to 16 hours for 72 m² (Cytec Industries Inc.,2015). In addition, external factors such as relative humidity also play a very important role in controlling void content in the lamina. In general, epoxy resin tends to absorb moisture in the air and trapped humidity is very difficult to remove under VBO processing. The effect of relative humidity on empty content of processed laminate VBO has been systematically studied (Nutt et.al 2010).

Method of inspection using ultrasonic C-scan was identified as a primary screening technique for determining the quality of an advanced composite material before it can be used in the field. The principles of ultrasound and disability carbon /

epoxy composite can occur during the fabrication process have been widely discussed (Brodal et al. 2011; Maeva et al. 2011; Shen & Dongri 2012; Shuck 2010; Venegas et al. 2012; Wright & Kingdom 2006; Zhangb et al. 2010) Among the defects that often occur in polymer-based composite materials are voids that exist in the laminate and the product during the manufacturing process. Defects such as voids and porosity often occurs due to the manner and appropriateness occur during the manufacturing process. Defects such as may result in a reduction in interlaminar shear strength (ILSS) of 5 % to 7 % for every 1 % increase voids are detected (AA 1997; Bowles & Frimpong 1992; Costa et al. 2001; Ghiorse 1993; H Jeong 1997; Schnars & Henrich 2006; Wisnom et al. 1996). With this reduction, it would lead to a reduction in strength as a whole. Generally, void exist in or between the layers of fibers used even in resin system itself. There are four mechanisms that can cause the formation of voids and porosity: (Loos & Springer 1983)

- (i) the air trapped during the accumulation layer.
- (ii) a slowdown in the resin.
- (iii) volatiles released from the curing process.
- (iv) internal pressure build-up of resin cure shrinkage.

### 1.2 Problem Statement

Generally, Cycom 5320 is a newly developed Carbon/Epoxy prepreg material by Cytec (Cycom 5320) for OOA application. Since this is a newly developed material, it required a determination of the most suitable bagging technique that produced high quality of composite laminate. In order to determine the quality of the laminate using pulse-echo ultrasonic C-scan, there is several problems have been highlighted as below:

- Porosity detection on the composite laminate especially on the laminate surface. Most of the quality inspection were conducted using non destructive and destructive method which is costly.
- ii. Insufficient solution of defect detection
- iii. Existing image and signal processing is not adequate to determine the overall composite quality
  - a. The problem of detection of localized porosity through the thickness in composite materials
  - b. Image processing: Most problem has been highlighted is the speckle region in the image. (Image noisy and blurred)
  - c. Signal processing : limitation of that highlighted is attenuation of ultrasonic waves is due to absorption and scattering phenomena especially for localized porosity.

Since the preparation of composite sample using prepreg material has been done in the environmental controlled room (controlled from dust, temperature and humidity), assumption has been made that there is no others defect affected sample during fabrication process except porosity/void defect.

### 1.3 Objectives.

The general objective of this research is to design, manufacture and test representative parts in order to evaluate and compare the capacities of autoclave and out-of autoclave manufacturing. The are the specific objectives that need to be achieved such as below:

- To produce carbon/epoxy composite panels using OOA technique by means of vacuum bagging and oven cure.
- ii. To compare the manufactured carbon/epoxy composite panels using UT (pulse echo method) with Matlab<sup>®</sup>.

iii. To analyse and correlate the porosity level using non destructive and destructive test.

### 1.4 Scope of Study

In this study, without any necessary tools, the non-destructive inspection of composite laminate will depend on the experiences in analysis the C-scan images. Most of the practice done will involve with destructive test to determine the defect, impurities or discontinuities in composite laminate such as determination of void content using burn off, acid digestion technique and mechanical test. In order to analyze the composite laminate quantitative and qualitatively, some of the researcher introduced data analysis algorithm. With this technique, comparison of signals and images been done. It also helps in eliminate noise signal. Matlab® will be used in order to analysis raw data from ultrasonic C-scan test. TOF and AMP data was analyzed accordingly in order to determine the thickness, porosity level and location. Besides that, density was measured in order to compare with the analyzed data. 3D images were produced and analyzed accordingly especially to determine the region that saturated with porosity. In this research, newly developed prepreg under tradename of Cycom 5320 was used as the main material which dedicated for OOA moulding and it has been produced by Cytec Industries Inc. As a comparison purpose Cycom 970 has been used. Cycom 970 is a prepreg that establishly used for autoclave moulding. Variation of bagging techniques were successfully conducted.

### **CHAPTER TWO**

### LITERATURE REVIEW

### 2.1 Out of Autoclave Technology (OOA)

The out of autoclave (OOA) or vacuum bag only (VBO) process depends largely on the extraction process or the physical and volatile air removal that aims to achieve low porosity which is time-dependent processes (C.Ridgard, 2016). The strategy to minimize the amount of curing period have been discussed in several scenarios by researchers and manufacturers especially involve in-tool and characterization of prepreg and adhesive film used which has freestanding postcure for monolithic and sandwich structure particularly after demoulding process (C.Ridgard, 2016).

Fibre placement known as the most popular fabrication route nowadays. R.W. Koon et. al (2016) investigated the fibre placement in order to investigate the quality of high rate automated fabrication for composite structure which is developed by Lockheed Martin Skunk Works<sup>®</sup> for MTM45<sup>®</sup>. They are comparing the quality of the product which is produced by autoclave and fibre placement technique. As finding, machine design and process parameters contributes a significant impact on improving the quality and cost for OOA fibre pacement process.

A.Haro et.al (2016) reported on rapid OOA composite manufacturing for aerospace grade prepregs which involved time-intensive and cost. Due to high demand for aerospace primary and secondary structure in the future with void free quality as well as performance wise, they introduced rapid process known as 'Quickstep' process (Figure 2.1) which involved with rapid heating and cooling with moderate compaction pressure. Two types of commercially available prepregs were utilized in the study for comparison among 'Quickstep', vacuum bag (VBO) and autoclave. Their system demonstrated more than 50 % total cycle time reduction

compared to VBO and autoclave with less that 1 % void content which is comparable with autoclave method.



Figure 2.1 : Quickstep QSE250 Machine and Curing Chamber (A.Haro et.al, 2016)

### 2.2 Ultrasonic C-Scan Testing

UT is the most popular nondestructive inspection method to examine the homogenous and heterogeneous materials. Frequency of 20 kHz to 20 MHz commonly applied in examining homogenous materials (i.e. noncomposite). When conducting composite materials evaluation, the operating frequency is usually below 5 MHz which is lower due to increment of attenuation inside materials and reduced the detection capability for tiny flaws. (Network & Guide 2013). Generally, UT can be used for varies material (metals, plastic and wood) in determining defect on surface and subsurface. Besides that, it also can be used in order to determine the thickness of materials via sound velocity and attenuation measurement (Krautkrämer & Krautkrämer 2013; Worlton 1956; Silk 1984; Willcox & Downes 2003; Birks et al. 1991)

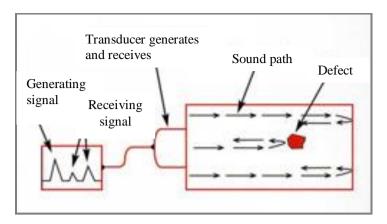


Figure 2.2: Ultrasonic testing principle (Krautkrämer & Krautkrämer 2013)

During inspection, high frequency pulses of sound wave are transmitted through the thickness of composite and sound reflecting time is measured. It is useful in determining void and delamination inside composite materials (Sfarra et al. 2013). Attenuation, signal scattering and absorption are the difficulties occurred in composite inspection especially in inspecting a thick and complex composite products.

### 2.2.1 Pulse Echo

Determination of the defected location in materials were measured using time required for UT pulse which generated by transducer to travel through the thickness of material and reflect back from defect location (i.e. voids, delamination and impurities) to transducer. Time interval is recorded as a few microsecond. There is two way transit time which is penetration and reflection the travel path. The relationship stated in Equation (i)

$$d = \frac{vt}{2} \text{ or } v = \frac{2d}{t}$$
 (i)

where [d] is the distance from the surface to the discontinuity in the test piece, v is the velocity of sound waves in the material, and t is the measured round-trip transit time.

### 2.2.2 Attenuation Measurement

Propagation of UT wave is influenced by the microstructure of the materials. Modulus of elasticity and density probably influence the UT wave velocity which normally governed by the phases and defect inside materials. Sum of absorption and scattering (UT attenuation) depends on damping and scattering from the grain boundary in the materials. Knowledge and data of thermophysical parameter is useful in order to characterize the produced attenuation signal which is difficult to quantify (Ru 2006) (Bastianini, Tommaso, et al. 2001).

The change of signal attenuation and conducting a simple qualitative measurement can be carried out which determine easy way compared to real measurement. This is known as relative measurement. Measurement of the relative attenuation measurements can be conducted by examining the exponential decay of multiple back surface reflections. Mechanical properties and variation in microstructure properties produced small changes in terms of attenuation and wave velocity (Bastianini, Tommaso, et al. 2001).

### 2.2.3 Defect Detection Using Ultrasonic C-Scan For Homogeneous Materials

Kapadia (2013) wrote about his work in deciding the utilization of frequency for imperfection location as a part of homogeneous materials. As a finding, the frequency regularly utilized is as a part of the scope of 20 kHz to 20 MHz (Network & Guide, 2013).

Summerscales (1990) reported they utilized of straight versatile break (linear elastic fracture) mechanics as a premise for the definition ought to be recognized to decide the measure of sub basic imperfections. It is because of the life of homogenous materials, for example, metal is controlled by the nucleation and development of splits or harm in the material (Summerscales 1990). Li et.al (2010) had concentrated on the examination of various NDT system which is thermography