

**MODELLING OF SOLAR IRRADIANCE FOR  
SIZING GLOBAL AND PERPETUAL SOLAR-  
POWERED UNMANNED AERIAL VEHICLE (UAV)**

**MUHAMMAD HAZIM BIN MASRAL**

**UNIVERSITI SAINS MALAYSIA**

**2016**

**MODELLING OF SOLAR IRRADIANCE FOR SIZING GLOBAL AND  
PERPETUAL SOLAR-POWERED UNMANNED AERIAL VEHICLE (UAV)**

**by**

**MUHAMMAD HAZIM BIN MASRAL**

**Thesis submitted in fulfilment of the  
Requirements for the degree  
of Master of Science**

**December 2016**

## ACKNOWLEDGEMENT

Alhamdulillah and praise to Allah the Al-Mighty as with His blessings, I was able to finish writing my thesis and my research. Without His guidance, I might not be able to complete writing this thesis on time. I am truly grateful that I have managed to finish it.

I want to express my deeply gratitude to my supervisor, Dr. Parvathy Rajendran for giving me endless guidance throughout the completion of this research. She has inspired me to make my own decision and to think outside of the box in order to do this project. Working with her even in this period has made me realize that no matter how high our academic level is, there is always room for more and for improvement because we are learning every day. Regarding this project, she has guided me on the step that I should take in order to achieve the objectives. Also, a big thanks to my co-supervisor Dr. Khairudin bin Mohamed, he has my utmost respect for being always humble in doing everything.

I also want to say thank you to all my friends who have helped me in doing this project. To Hairuniza and Zaim who have provided me their assistance during the analysis process of using EUREQA and MATLAB in order to find out the best model. Thank you very much, without your help I would not be able to do it on my own.

Again, I am truly grateful that this research is completed. I hope everything will end well for everyone. Thank You.

## TABLES OF CONTENTS

	<b>Page</b>
<b>ACKNOWLEDGEMENT</b>	<b>ii</b>
<b>TABLES OF CONTENTS</b>	<b>iii</b>
<b>LIST OF TABLES</b>	<b>vi</b>
<b>LIST OF FIGURES</b>	<b>vii</b>
<b>LIST OF SYMBOLS</b>	<b>ix</b>
<b>LIST OF ABBREVIATIONS</b>	<b>xii</b>
<b>ABSTRAK</b>	<b>xiv</b>
<b>ABSTRACT</b>	<b>xvi</b>
<b>CHAPTER ONE: INTRODUCTION</b>	
1.1 Overview	1
1.2 Research Background	1
1.3 Solar Irradiance	3
1.3.1 Direct Solar - Ground Measurements	4
1.3.2 Indirect Solar - Estimation using Satellite Observation	6
1.3.3 Solar Irradiance Modelling	7
1.4 Global Solar-powered UAV Operation	8
1.5 Problem Statements	10
1.6 Research Objective	10
1.7 Research Scope	10
1.8 Thesis Outline	12

## **CHAPTER TWO: LITERATURE REVIEW**

2.1	Overview	14
2.2	Review of the Existing Solar Irradiance Models	14
2.3	Review of the Global and Perpetual Solar-powered UAV	23
2.3.1	Existing Study for Perpetual UAV	23
2.3.2	Existing Solar-powered UAV	25
2.3.3	Existing Study for Global Operation	31

## **CHAPTER THREE: METHODOLOGY**

3.1	Overview	34
3.2	Solar Irradiance Modelling	34
3.2.1	Local Solar Model	35
3.2.2	Global Solar Model	44
3.2.3	Global and Perpetual Solar-powered UAV Sizing	51
3.2.4	Implicated Parameters	52
3.2.5	Solar-powered UAV Design Model	52

## **CHAPTER FOUR: RESULTS AND DISCUSSION**

4.1	Overview	57
4.2	Local Solar Simulations	57
4.3	Global Solar Simulation	69
4.4	Solar Model Validation	73
4.5	Perpetual UAV Optimization	75

## **CHAPTER FIVE: CONCLUSION**

5.1	Overview	83
5.2	Research Summary	83
5.3	Future Work	85

<b>REFERENCES</b>	<b>86</b>
-------------------	-----------

## **APPENDICES**

Appendix A: Existing UAVs with various power source

Appendix B: Tabulate Parameter for each model

## **LIST OF PUBLICATIONS**

## LIST OF TABLES

		<b>Page</b>
Table 1.1	Advantages and disadvantages for both Satellite and UAV	9
Table 1.2	Cities study for UAV	12
Table 3.1	Geographical location of place of interest for present study	42
Table 3.2	Coordinates for all cities chosen for global simulation	45
Table 3.3	Fix value for UAV input parameter	55
Table 4.1	The list of all parameters, its abbreviation and units	62
Table 4.2	The list of proposed model and its parameters	62
Table 4.3	Comparison of Proposed Solar Irradiance Model with Existing Model	68
Table 4.4	Table of yearly average solar irradiance and daylight duration for 2013	69
Table 4.5	Comparison of monthly average solar irradiance for various value of	74
Table 4.6	Simulation and sizing comparison between AUAV and GUAV design	79

## LIST OF FIGURES

		<b>Page</b>
Figure 1.1	Schematic of a pyranometer with shading disk.(Instruments, 2009)	4
Figure 1.2	Schematic of a pyranometer with shading disk (Instruments, 2009)	5
Figure 2.1	Sunrise 1.(GENUTH, 2015)	26
Figure 2.2	Gossamer Penguin.(Maccready, Lissaman, Morgan, & Burke, 1983)	27
Figure 2.3	Solar challenger. ("Maccready Solar Challenger," 2014).	28
Figure 2.4	Sun seeker.(Moore, 2004)	29
Figure 2.5	Pathfinder.("Pathfinder Solar-Powered Aircraft," 2013).	29
Figure 2.6	Solar impulse (Emigepa, 2015).	30
Figure 3.1	Station Network of Malaysian Meteorological Department (mosti, 2016).	38
Figure 3.2	Layout of five meteorological station location with observe solar irradiance data (shutterstock, 2003)	42
Figure 3.3	Local Solar Model Flow Chart.	43
Figure 3.4	All twelve selected cities on a world grid map	45
Figure 3.5	The framework of solar irradiance and daylight duration modelling (P. Rajendran& Smith, 2016)	46
Figure 3.6	Earth-sun vector declination angle on June 22(Parvathy Rajendran, Smith, & bin Masral, 2014)	50
Figure 3.7	Solar geometry of a sloped surface(Parvathy Rajendran et al., 2014)	50
Figure 3.8	The Solar-powered UAV Design Model (Parvathy Rajendran& Smith, 2015b)	53
Figure 3.9	Model development flow chart	56
Figure 4.1	Humidity Average for Five Cities	58



Figure 4.2	Cloud Cover Average for Five Cities	58
Figure 4.3	Pressure Average for Five Cities	59
Figure 4.4	Gust Speed Average for Five Cities	59
Figure 4.5	Wind Speed Average for Five Cities	60
Figure 4.6	Rain Precipitate Average for Five Cities	60
Figure 4.7	Temperature Average Data for Five Cities	61
Figure 4.8	The actual solar irradiance and the estimated using PM12.	67
Figure 4.9	Yearly average daylight duration for 2013	70
Figure 4.10	Yearly average solar irradiance for 2013	71
Figure 4.11	Solar irradiance against day of the year	72
Figure 4.12	Daylight duration against day of the year	72
Figure 4.13	Solar irradiance for Kuala Lumpur on 2013	75
Figure 4.14	The wingspan of AUAV	76
Figure 4.15	The maximum take-off weight of AUAV	77
Figure 4.16	The solar module to wing area ratio for AUAV	78
Figure 4.17	Endurance of AUAV for various cities in 2013	80
Figure 4.18	Endurance of GUAV for various cities in 2013	81
Figure 4.19	Perpetual flight for AUAV and GUAV design feasible over a year.	81

## LIST OF SYMBOLS

$n$	Hours of bright sunshine
$N$	Hours of possible sunshine
$l_o$	Solar constant
$H_{max}$	Maximum radiation
$\varphi$	Latitude
$\gamma$	Longitude
$\alpha$	Altitude
$n_{di}$	Number of day
$DN$	Number of day (precise)
$n_d$	Day of each month
$\delta$	Declination angle
$\omega_s$	Sunrise hour
$\omega$	Hour angle
$s/s_o$	Relative number of sunshine hours
$R_j$	Transmission function for hourly variation
$R_i$	Transmission function for daily variation
$H_{ext}$	Extraterrestrial solar irradiance
$H$	Global solar irradiance
$H_{bh}$	Beam radiation on horizontal plate
$H_d$	Diffuse solar irradiance
$\delta_o$	Total optical depth
$\varepsilon$	Eccentricity correction
$Z$	Zenith angle

$\theta_{zt}$	Zenith angle at certain time
$m$	Air mass
M	Month
<i>GMT</i>	Green meridian time
<i>EOT</i>	Equation of time
<i>AST</i>	Apparent solar time
<i>SOLALT</i>	Solar altitude
<i>SOLAZM</i>	Solar azimuth
<i>SOLINC</i>	Solar incidence angle
<i>TILT</i>	Tilt angle
<i>WAZM</i>	Wall azimuth angle
<i>SOLHRA</i>	Solar hour angle
$P_{Solar}$	Solar module system power (W)
$Ir_{Max}$	Solar irradiance (W/m <sup>2</sup> )
$eff_{Solar}$	Efficiency of solar module
$eff_{MPPT}$	Efficiency of maximum power point tracker (MPPT)
$A_{Solar}$	Solar module area (m <sup>2</sup> )
$P_{Required}$	UAV power required (W)
$C_L$	Lift coefficient
$C_D$	Drag coefficient
$W_{TOmax}$	Maximum take-off weight (N)
$P$	Air density (kg/m <sup>3</sup> )
$S$	Wing area (m <sup>2</sup> )
$V$	Air speed (m/s <sup>2</sup> )
$C_{Do\_W}$	Zero-lift-drag coefficient

$E$	Oswald efficiency
$AR$	Wing aspect ratio
$B$	Wing span (m)
$W_{Struct}$	Structure weight (N)
$W_{Batt}$	Battery weight (N)
$W_{Solar}$	Solar power system weight (N)
$W_{Electric}$	Propulsion system weight (N)
$W_{Ctrl}$	Control system weight (N)
$W_{Pay}$	Payload weight (N)
$\overline{H_{obs}}$	Averaged observed solar irradiance
$H_{obs}$	Observed solar irradiance
$H_{est}$	Estimated solar irradiance

## LIST OF ABBREVIATIONS

PM	Proposed model
<i>RMSE</i>	Root mean square error
$R^2$	Coefficient of determination
<i>MBE</i>	Mean bias error
MSE	Mean squared error
MTOW	Maximum take-off weight
SI	Solar irradiance
UAV	Unmanned aerial vehicle
AUAV	Average Unmanned aerial vehicle
GUAV	Global Unmanned aerial vehicle
RE	Renewable energy
PV	Photovoltaic
NASA	National Aeronautics of space administration
GSI	Global solar irradiance
GTI	Global tilted irradiance
DHI	Diffuse horizontal irradiance
GHI	Global horizontal irradiance
DNI	Direct normal irradiance
FOV	Field of view
GOES	Geostationary operational environmental satellite
API	Air pollution index
ANN	Artificial neural network
PM	Particulate matter

PEMFC	Polymer electrolyte membrane
DFMC	Direct methanol fuel cells
RAT	Ram air turbine
SoDa	Solar radiation data
MMD	Malaysia meteorology department
TSP	Total suspended particulate
MOSTI	Ministry of science, technology and innovation
ANN	Artificial Neural Network
BP	Back Propagation

**PEMODELAN BAGI SINARAN SURIA UNTUK PENSAIZAN GLOBAL DAN  
PENERBANGAN BERTERUSAN BAGI PESAWAT SOLAR TANPA PEMANDU**

**ABSTRAK**

Sejak kebelakangan ini terdapat pelbagai aplikasi kenderaan kecil udara tanpa pemandu untuk kedua-dua penerbangan tentera dan awam. Peluang untuk meningkatkan keupayaan UAV berkuasa solar untuk beroperasi seperti satelit sebagai pseudolite. Walau bagaimanapun, berat dan sekatan kuasa sentiasa menjadi punca utama yang menghadkan UAV untuk beroperasi sebagai menyerupai satelit. Tujuan kajian ini adalah untuk merapatkan jurang ilmu di dalam bidang penerbangan berterusan di seluruh rantau di seluruh dunia untuk operasi selama setahun. Melalui kajian ini, tumpuan telah diberikan 1) untuk mengkaji model solar tempatan dan global serta 2) bagaimana data solar mungkin membolehkan kemungkinan membangunkan reka bentuk UAV bagi operasi global. Dalam kajian ini, untuk pemodelan solar tempatan, 14 model telah diusul dan dikaji dengan parameter yang berkaitan menggunakan teknik regresi. Untuk pemodelan sinaran global, pendekatan secara teoritikal digunakan untuk mengkaji kesemua dua belas bandar. Prestasi setiap model solar yang dicadangkan dalam menganggarkan sinaran suria untuk Malaysia dianalisa berdasarkan Ralat Relatif Min Square (*RMSE*), Pekali Penentu ( $R^2$ ) dan Min Ralat Pincangan (*MBE*). Analisis secara simulasi telah dilaksanakan bagi memahami kemampuan untuk penerbangan kekal bagi pesawat tanpa pemandu. Di sini, terdapat 2 jenis kajian 1) AUAV; simulasi data solar tahunan purata untuk reka bentuk UAV solar yang sesuai untuk operasi di setiap bandar-bandar, dan 2) GUAV; satu UAV yang dioptimumkan reka bentuknya yang mampu untuk beroperasi di semua bandar. Model

yang ke 12 telah di cadangkan sebagai model bagi Malaysia, dengan penemuan *RMSE* 0.856%, *MBE* 0.282% dan  $R^2$  0.988. Spesifikasi untuk pesawat tanpa pemandu global yang telah disimulasi sepanjang tahun adalah, 0.5057 kg bagi Berat Berlepas Maksima, Panjang Kepak 2.03 m dan Nisbah Kawasan Sayap sekitar 40%. Berdasarkan sinaran suria global bagi tahun 2013, kami berjaya mengenal pasti faktor resapan bagi Malaysia adalah 60.8% dengan menggunakan teknik interpolasi.

*Kata kunci*; radiasi solar global; data jabatan meterologi; awan; suhu; kelajuan angin; kelajuan tiupan; hujan; lembapan; tekanan; kaedah regresi; kapal terbang tanpa pemandu; beban; konfigurasi saiz



# MODELLING OF SOLAR IRRADIANCE FOR SIZING GLOBAL AND PERPETUAL SOLAR-POWERED UNMANNED AERIAL VEHICLE (UAV)

## ABSTRACT

There are various applications of small unmanned aerial vehicle for both military and civil aviation in recent years. The opportunity to enhance of solar-powered UAV capabilities to operate as pseudolite satellite enables all countries around the world an endless possibility of their own technology. However, weight and power restriction are always the main cause which limits UAV to operate as pseudolite satellite. The aim of this work is to bridge the knowledge gap in the area of perpetual flight across regions around the world for a yearlong operation. Through this study, focus has been given 1) to study both local and global solar modelling and 2) how these solar data may enable the possibilities of developing a UAV design capable of global operation. In this study, for local solar modelling, 14 variant proposed models are studied with its relevant parameters using the regression technique. Then, as for global solar modelling, the theoretical approach has been used to study twelve cities spreading all over the world. The performance of each proposed solar model in estimating the solar irradiance (SI) for Malaysia is analysed based on its Root Mean Square Error (*RMSE*), Coefficient of Determination ( $R^2$ ) and Mean Bias Error (*MBE*). Finally, the analyses carried out to understand the feasibility of a perpetual solar-powered UAV operated around the world have been simulated. Here, the two modes of study include 1) AUAV; the yearly average solar data simulation to design solar-powered UAV suitable for operation at each city, and 2) GUAV; a single optimized solar-powered UAV design that is capable for operation in all cities investigated in this work.

Proposed Model (PM) 12 had been recommended as solar estimation model for Malaysia, the result showed that the *RMSE* for the model is 0.856%, *MBE* 0.282% and  $R^2$  is 0.988. An optimum design specification for Global UAV (GUAV) successfully to simulate perpetual flight for whole year with specification of Maximum Take-Off Weight (MTOW) 0.5057 kg, Wingspan (S) 2.03 m and Solar Wing Area Ratio (SWAR) 40%. Based on Global SI data year 2013, we able to estimate the best diffuse factor value for Malaysia, which is 60.8% by using interpolation method.

*Keywords-component*; global solar irradiance; meteorology station data; clouds cover; temperature; wind speed; gust speed; rain precipitate; relative humidity, pressure, regression method; UAV; pseudolite; weight; sizing configuration.

# **CHAPTER ONE**

## **INTRODUCTION**

### **1.1 Overview**

For this chapter, 1.2 and 1.3 will discuss the available renewable energy, type of solar irradiance and the equipment that are used to measured solar radiation. The objective, problem statements and the research scope are presented in 1.5 till 1.7. Thesis outline will explain more about chapter overview for each of it.

### **1.2 Research Background**

Energy is important to drive this world's functionality. For instance, energy is divided to two categories, namely renewable and unrenewable energy. Renewable energy (RE) is clean and sustainable energy which has good potential to reduce the greenhouse effect and it is also a good substitute to the depleting fossil fuels (Mirzaei,Tangang et al., 2014). Study on renewable energy is crucial since fossil fuels are still the primary energy supply which is almost 88%. Based on data in 2008, the present oil supply will only last for 42 years. Hence, in less than half a century, our crude oil supply may be depleted (Urguhart, 2011).

In addition, the development and study on RE application have been diversified in the past few years, earning its high appreciation (Solangi,Lwin et al., 2011). Some of the different types of RE resources which may be utilized to produce clean energy are solar, hydro, wind, wave, thermal and biomass. The two most common technologies of solar are the photovoltaic (PV) cells and solar thermal energy. PV cells convert solar radiation to electricity, whereas solar thermal energy implements the concept of heating and cooling through absorption and emission of solar radiation.

Ironically, the study on solar power system has been around for more than 100 years, till the discovery of the PV cell. PV evolution increases drastically since; from 1 to 2 % efficiency around 1876, up to 10 % efficiency in late 1970. Since then, the world record for solar cell efficient has achieved almost 44.7 % in efficiency (Daily, 2013) developed by Fraunhofer Institute for Solar Energy Systems. Besides, technology has advanced much with a variety of rigid, semi-rigid and flexible PV development and available commercially-off-the-shelf.

Subsequently, many countries in the world have already implemented and developed their solar power plants as an alternative power generator (Urguhart, 2011). The total amount of power produced at present solar power plant in Malaysia is about 20,493MW. Moreover, the power consumption is estimated to increase up to 23,099MW in 2020, whereas thermal power plants and hydro power plants in Malaysia produced 7,103MW and 1,911MW respectively for year 2013. Therefore, generating energy source through solar is a good option as it provides unlimited renewable energy. In addition, it does not pollute the surrounding or produce any hazardous waste(Hazim Masral,Parvathy Rajendran\* et al., 2015).

There is an extensive use of solar energy especially in industries such as military, telecommunication, agricultural, water desalination, aviation and building industry. Common applications of solar power systems includes pumps, engines, fans, refrigerators and water heaters. Lately, both the military and aviation industries have raised their initiatives in solar energy development in order to reduce its dependence on fossil fuels(Mekhilef,Saidur et al., 2011).

It is believed the military will play a key role in leading and bringing solar to the mass market. Therefore, this study intends to study the feasibility of conventional solar-powered Unmanned Aerial Vehicle (UAV) global solar-powered flight operations. Currently, the existing solar-powered UAV are Solar Eagle, Zephyr,

Helios, Sky-Sailor and Solong (Smith&Rajendran, 2014). As to date, a lot of efforts has been done to establish long endurance UAV over the past few decades.

Thus, this study aims to understand 1) the meteorology parameter and its relation in solar irradiance intensity and 2) to study feasibility of a small solar-powered UAV for global operation. Besides, the knowledge on available solar irradiance is essential in various application, especially to design highly efficient solar module system, which solely depends on the amount of solar radiated to earth surface. Moreover, the study for global operation may elucidate the practicality of solar-powered flight in aviation industry.

### **1.3 Solar Irradiance**

Generally, solar radiations are divided into two types, 1) solar radiation outside the earth atmosphere or known as extra-terrestrial solar irradiance and 2) solar radiation at the earth surface. The solar radiation on earth surface part consists of three main elements, which are global, direct and diffuse solar irradiance. The global is the combination of direct and diffuse solar irradiance. The direct solar irradiance is the amount of radiation that goes through atmosphere falling in a unit area perpendicular to the beam at Earth's surface.

However, the diffuse solar irradiance is the solar irradiance that is scattered and reflected through the cloud, air particles and other atmospheric components that are available in the sky that may prevent direct solar radiation (Zhao,Zeng et al., 2013, Kaushika,Tomar et al., 2014, Mohammadi&Khorasanizadeh, 2015) refer Figure 1.1.

Information about solar data availability is vital in the assessment and modelling of active solar energy systems, regardless if it is for aircraft or home application. The main two parameters involved in this study are the solar irradiance intensity and solar irradiance availability duration or commonly referred to daylight

duration. Conventionally, solar irradiance data collection can be obtained in three techniques; 1) direct collection through in-situ ground measurements, 2) indirect methods through satellite estimation and 3) modelling techniques.

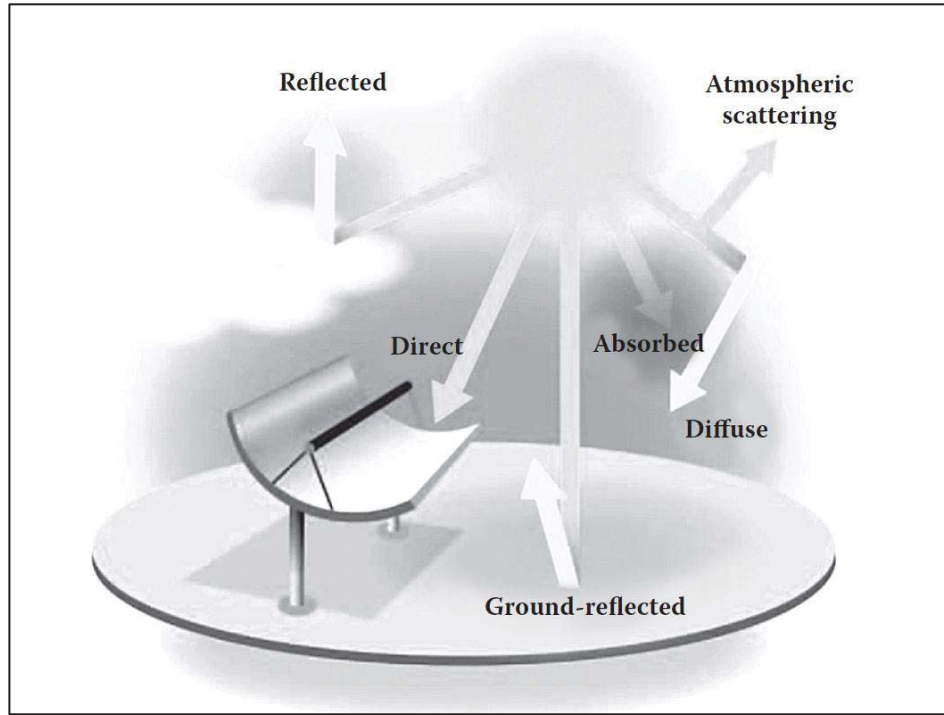


Figure 1.1 Solar Irradiation components

### 1.3.1 Direct Solar - Ground Measurements

In-situ measurement is more complex as compared to indirect method, as it requires measurement sensor and equipment that is highly cost as well as their maintenance. Meanwhile, to ensure reading from an instrument is consistent with other measurements, every equipment requires yearly calibration to establish the reliability of the instrument. The equipment that is normally used for measuring and monitoring solar radiation is pyranometers.

Other than that, several other equipment and sensors such as hygrometer, anemometer, barometer, thermometer and rain gauge to measure typical weather parameter. Unfortunately, this instrument is not widely used in this country, only certain meteorology stations are available with pyranometer and other stations not.

Due to this constraint, only a few data locations are available for the solar irradiance study. In addition, most of the available data worldwide consisting of extra-terrestrial solar irradiance are either limited or without diffuse irradiance component.

Generally, Pyranometers as shown in

Figure 1.2 are used to measure total hemispherical radiation or horizontal solar irradiance for horizontal surface. The incoming solar irradiance from a  $2\pi$  solid angle can be measured on a planar surface,  $180^\circ$  from horizon to zenith to horizon and  $360^\circ$  around the horizon. It consists of an outer glass dome and inner glass dome covers made of glass for shield the electrical equipment such as sensor from thermal convection. The dome also functions to protect equipment from weather (rain, dust and wind). A cartridge or desiccator contains silica gel inside the dome for soaking up water vapor and maintaining the dome dry. Moreover, pyranometer also measures the tilted irradiance for tilted surface. The diffuse solar irradiance (DHI) can also be measured using the pyranometer, while the direct beam component will be eliminated by small shading disk. The pyranometer needs to continuously be shaded, so the disk will be mounted on an automated solar tracker. Hence, the disk acts as a shadow ring and may prevent the direct component (DNI) from reaching the sensor as shown in Figure 1.2.

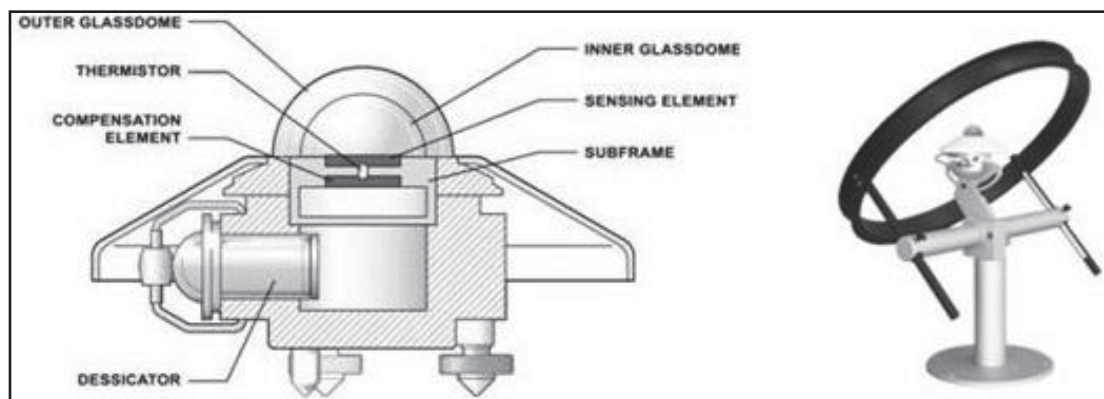


Figure 1.2: Schematic of a pyranometer with shading disk.(Instruments, 2009)

### **1.3.2 Indirect Solar -Estimation using Satellite Observation**

The assessment of the solar energy resources over large areas and the elaboration of solar radiation maps are nowadays conducted using satellite derived information. Among the existing methodologies, those based on parametric models are common to estimate the clear-sky solar radiation. However, the fittingness of that option depends both on the accuracy of the solar radiation algorithm and the satellite derived information and their relationship (López&Batlles, 2014). However, it is very expensive and difficult to maintain as it can only be sustained through communication between satellite and ground stations. Even though, there are free data available of the interest location from National Aeronautics and Space Administration (NASA), the data are out dated for most locations (NASA, 2013).

In the last decades, various models have been produced for estimating solar irradiance using geostationary satellite image (Blanc,Gschwind et al., 2011, Marquez,Pedro et al., 2013, Rusen,Hammer et al., 2013). Geostationary satellites are useful for various applications and mission, and one of it is for meteorological purposes. The images that are captured by the meteorological satellite over a large area and with high quality also allow recognition and estimation of the clouds' change. These data may be further analysed to generate forecast of solar radiation at the ground level.

There are several weather satellites still in operation, such as the Geostationary Operational Environment Satellite system GOES-12, GOES-13, and GOES 15 which are owned by United States. The Elektro-L 1 is a new-generation weather satellite by Russia (Zak, 2013) and the MTSAT-2 located over the mid Pacific is owned by Japanese (JMA, 2014). The European have the Meteosat series from 6 till 9, which covers Indian and Atlantic Ocean. Besides, China currently has three Fengyun series